NASA TECHNICAL MEMORANDUM



NASA TM X-3406

SUBJECTIVE RESPONSE TO COMBINED NOISE AND VIBRATION DURING FLIGHT OF A LARGE TWIN-JET AIRPLANE

Sherman A. Clevenson

Langley Research Center

Hampton, Va. 23665



1. Report No.	2. Government Accession No.	3. Recipient's Catalog No.
NASA TM_X-3406		
4. Title and Subtitle		5. Report Date
SUBJECTIVE RESPONSE TO	COMBINED NOISE	October 1976
AND VIBRATION DURING FI	IGHT OF A LARGE	6. Performing Organization Code
TWIN-JET AIRPLANE		
7. Author(s)		8. Performing Organization Report No.
Sherman A. Clevenson		L-10578
		10. Work Unit No.
9. Performing Organization Name and Address		504-09-21-01
NASA Langley Research Cent	er	11. Contract or Grant No.
Hampton, VA 23665		The definition of the first
		13. Type of Report and Period Covered
12. Sponsoring Agency Name and Address		Technical Memorandum
National Aeronautics and Spa	ce Administration	14. Sponsoring Agency Code
Washington, DC 20546		14. Sponsoring Agency Code
15. Supplementary Notes		

16. Abstract

This report presents the results of a study in which an NASA twin-jet airplane was used to obtain controlled noise and vibration environments during flight while obtaining subjective responses from 13 passenger-subjects (6 females and 7 males). Subjective ratings of overall comfort, comfort when considering only vibration, and comfort when considering only noise were obtained during times of different vibration and noise environments. Passenger-subjects were able to distinguish and rate noise better than vibration in this study. In addition, there was a statistically significant difference in ratings of ride comfort due to both sex type and flight experience. Males rated flying discomfort much more severely than females when rating the overall ride and the ride when considering only the noise environment. Experienced passengers also rated the overall ride to be more uncomfortable than inexperienced passengers.

17. Key Words (Suggested by Author(s))	18. Distribution Statement			
Passenger response Transport aircraft Ride quality Noise and vibration en Comfort criteria	vironment	Un	classified — Unl	imited Subject Category	53
19. Security Classif. (of this report) Unclassified	20. Security Classif. (of this Unclassified	page)	21. No. of Pages 41	22. Price* \$ 3.75	

SUBJECTIVE RESPONSE TO COMBINED NOISE AND VIBRATION DURING FLIGHT OF A LARGE TWIN-JET AIRPLANE

Sherman A. Clevenson Langley Research Center

SUMMARY

Thirteen passenger-subjects were asked to rate their comfort associated with noise and vibration during a 1-hour flight aboard an NASA twin-jet airplane. The airplane was flown in a straight, horizontal path at a 3962-meter (13 000-foot) altitude. A range of vibration and noise levels was obtained by varying the thrust, the forward speed, and the position of the landing gear and drag brakes. The subjects were asked to rate designated 1-minute segments of the flight. The passenger-subjects' ages ranged from 18 to 48 years, weight from 54 to 84 kg (118 to 185 lb), and flight experience from first flight to many flights. Seven subjects were males and six were females; nine were professionals, three were students, and one was a homemaker.

For this flight, the noise and vibration levels were essentially independent of each other. For constant vibration level, good correlations were obtained between the vibration ratings and noise ratings, vibration ratings and noise levels, and noise ratings and noise levels. However, at constant noise levels, although the noise rating correlated well with vibration rating, both the noise rating and the vibration rating showed poor correlations with vibration level. Passenger-subjects were able to distinguish and rate noise better than vibration.

There was a statistically significant difference in ratings of ride comfort due to both sex type and experience in flying as passengers. Males rated flying discomfort much more severely than females when rating the overall ride and the ride when considering only the noise environment. Experienced passengers rated the overall ride more severely than inexperienced passengers.

INTRODUCTION

The problem of vibration and audible noise in aircraft has always affected the feeling of well-being, that is, comfort or discomfort, while flying. The solution to the problem is compounded by both the paucity of in-flight vibration and noise measurements and by the lack of subjective measurements with a controlled environment. Although few data on subjective responses are published for actual aircraft flights (see, for example, ref. 1), some

studies have been made by using airborne or ground-based simulators. (For example, see refs. 2, 3, and 4.) A bibliography on ride-comfort studies is published in a recent AGARD paper (ref. 5), and a number of recent studies are reported in reference 6. One study conducted by the University of Virginia used paying passengers in commercial aircraft (ref. 1). However, only limited questionnaires could be used with these passengers and there was no control of the noise or vibration of the aircraft. A second University of Virginia study (ref. 2) utilized the NASA Jetstar airplane with the GPAS (General Purpose Airborne Simulator) system in which only two test subjects could be flown at one time. The ground-based simulator tests (refs. 3 and 4) utilized from two to six subjects in a controlled vibration environment, but the feeling of flying at an altitude of thousands of meters in an airplane could not be simulated. The current flight study provided realism by utilizing a commercial-type twin-jet airplane; the noise and vibration environment was varied by changes in the airplane configuration.

In the course of using ground-based simulators, investigators have looked for effects of sex type and flight experience on comfort ratings. Reference 4 showed that sex type had negligible effect, whereas data from reference 7 showed that males rated more severely than females in the rms (root-mean-square) vibration range below 0.04g; that is, the same ride was rated more uncomfortable by males. The effect of flight experience was shown to be negligible in references 3 and 7. The subjects selected for the current study included both males and females as well as experienced and inexperienced passenger-subjects.

The current study consisted of recording subjective responses of a group of males and females while flying in the NASA twin-jet airplane during the various noise and vibration conditions created while the airplane was in straight and level flight. For each condition the noise and vibration environment was also recorded. The main objective of this study was to obtain and correlate flight objective data and subjective data, from which the effect of noise and vibration on human-comfort response would be determined. Secondary objectives were to study the effects of sex type and flight experience on the response rating of the passengers.

EXPERIMENTAL DESIGN

The NASA twin-jet airplane (fig. 1) was used for obtaining flight objective and subjective measurements. Some of the physical characteristics of the airplane are listed in table I. Vibration and noise measurements were recorded throughout the flight. Subjective responses were obtained during 13 1-minute segments during level flight at a 3962-meter (13 000-foot) altitude. The weather was clear with no apparent clear-air turbulence. The noise and vibration environment was varied by changing engine thrust, wheel position (retracted or extended), drag-brake position, and airspeed. The 13 passenger-

subjects consisted of 7 males and 6 females whose ages ranged from 18 to 48 years and had flight experience ranging from first flight to having flown over 160 934 km (100 000 miles).

After each segment of flight had stabilized (1 to 2 minutes), the passenger-subjects were asked to consider 1 minute of flight and then to complete the three-part questionnaire shown in table II. The first part consisted of rating the overall ride by utilizing the adjectives "very comfortable," "comfortable," "acceptable," "uncomfortable," and "very uncomfortable." For later analysis, these adjectives were replaced by numbers 1, 2, 3, 4, and 5, respectively, and were called overall ratings R_0 . The other two parts of the questionnaire concerned rating the noise and vibration in terms of comfortable and uncomfortable on a 5-point scale where 1 was at the most comfortable end of the scale and 5 was at the most uncomfortable end of the scale. These noise comfort and vibration comfort ratings were designated by R_n and R_v , respectively.

After completion of the 13 flight segments, each passenger-subject was requested to complete a second questionnaire shown in table III. The purpose of this questionnaire was to obtain both demographic data on the passenger-subjects and to obtain their reactions to various ideas and conditions. A summary of the responses to this questionnaire is given in appendix A.

The seating locations of the passenger-subjects and the placement of the instrumentation are shown in figure 2. The passenger-subjects were located both behind and in front of the trailing edge of the wing, and the vibration-measuring equipment was located near the center of gravity of the airplane. All seats in this airplane are shown in figure 2; the airplane was not configured as the typical commercial vehicle with reference to the number of seats, but the rest of the interior appeared as a commercial airplane in regard to paneling, lighting, and other features.

The instrumentation consisted of measuring and recording equipment for vibration, noise, and temperature. Vibrations were measured by using accelerometers powered by batteries, with all being enclosed in portable containers as shown in figure 3. The outputs of the accelerometers were recorded on six FM channels of a seven-channel portable flight tape recorder. The frequency response of the vibration-measuring system was from 0 to 25 Hz. Two sets of three orthogonal accelerometers were used. Additional details of this system are given in reference 8. The noise was measured on a sound-level meter and was recorded directly on the seventh channel of the tape recorder. The temperature of the airplane was measured with a laboratory thermometer and was manually recorded on a data sheet. The temperature was maintained essentially constant at about 24° C (75° F) by the air-conditioning system of the airplane.

RESULTS AND DISCUSSION

The results of this investigation are shown both graphically and in tabular form. The measured noise and vibration environments are given in table IV and shown in figure 4. The averages for the ratings $R_{\rm O}$, $R_{\rm V}$, and $R_{\rm n}$ for all subjects are given in table IV; for males and females, in table V; and for experienced and inexperienced passenger-subjects, in table VI. These data are shown in figures 5 to 12. Typical standard deviations of subjective ratings for a given test condition were of the order of ± 0.9 .

Effect of Noise and Vibration

To study the individual effect of noise and vibration, it is first necessary to show that little or no correlation exists between noise and vibration. Thus, if the vibration levels always increased as the noise increased, there would be a high degree of correlation and it would be impossible to correlate a rating with one and not the other. The measured noise and vibration data from the flight are shown in figure 4. Since the measured lateral vibrations were considerably less than the measured vertical vibrations, all comparisons will be made by considering only vertical vibrations in this paper. It can be seen that little correlation exists between noise and vibration during this flight. Thus, it may be possible to show which environment affects the ratings, or at least to show how the passenger-subjects distinguish and rate the noise and vibration environments.

To study the individual effects, the data of figure 4 were separated into groups of essentially constant noise levels (in dB(A)) and rms vibration levels (in g units). The "constant" group vibration levels to be considered are $g_{rms} = 0.0344 \pm 0.0059$, 0.0498 ± 0.0051 , and 0.1095 ± 0.0066 indicated by acceleration groupings a_1 , a_2 , and a_3 , respectively, in figure 4. The "constant" group noise levels to be considered are 82.5 ± 1.5 dB(A), 88 ± 2 dB(A), and 93 ± 2 dB(A) indicated by the noise groupings n_1 , n_2 , and n_3 , respectively, in figure 4.

The ratings due to vibration are shown as a function of measured noise environment (in dB(A)) in figure 6. Collectively, the data show a correlation coefficient r = 0.622 between vibration rating and noise level. At the medium and high vibration levels, an

increase in noise shows an increase in vibration ratings. (Number 5 represents maximum discomfort.) At the low vibration level, group a_1 , there appears to be no trend of vibration rating compared to the noise environment. Based on the higher vibration levels, groups a_2 and a_3 , it may be concluded that the noise environment does affect the vibration rating.

The ratings due to noise R_n as a function of noise level (in dB(A)) while holding acceleration constant are shown in figure 7. The data have a high correlation coefficient (r = 0.715), and at each vibration level, the trend of increased numerical rating with increased noise level is shown; thus, this noise environment is an excellent indicator of comfort due to the noise environment.

To observe the effects of vibration, the data are replotted for three "constant" noise levels. The noise rating is shown as a function of vibration rating in figure 8. The correlation coefficient is the same (r=0.854) as in figure 5. (The same data are plotted differently.) The vibration rating $R_{\rm V}$ is shown as a function of rms acceleration level in g units in figure 9. There is little correlation between $R_{\rm V}$ and acceleration level (r=-0.401). The data of figure 9 indicate that passenger-subjects do not sort out vibration or rate comfort due to vibration very well on this airplane. Some other factor, specifically the noise environment as shown in an earlier figure, is causing or interacting with the comfort rating due to vibration.

The noise comfort rating R_n is shown as a function of vibration (acceleration) level for constant noise levels in figure 10. As might be expected, little correlation exists between noise rating and vibration level (r = -0.219).

In order to visualize the effects of noise and vibration better, the data for the center groupings of vibration and noise levels have been extracted from figures 5 to 10 and are shown in figure 11. The correlation coefficients (r) for each set of data are shown above each plot. Note the high correlation coefficients for both $R_{\rm V}$ and $R_{\rm n}$ as functions of noise level for constant $g_{\rm rms}$. In the lower half of the figure, note the very low (one is negative) coefficients for $R_{\rm n}$ and $R_{\rm v}$ as functions of vibration level where the noise is constant. Thus, although $R_{\rm v}$ and $R_{\rm n}$ are correlated to each other, they correlate well only with noise environment and not with vibration environment on this airplane. It is apparent that the passenger-subjects can distinguish and rate noise better than they can distinguish and rate vibration in this airplane.

A comparison of how males and females rate comfort in this airplane is indicated in table V and figure 12. The ratings $R_{\rm n}$, $R_{\rm v}$, and $R_{\rm o}$ are compared between male and female passengers on this flight. The 45° line on each plot divides the ratings as follows: If all data fall on this line, the males and females are rating the flight segments the same; that is, they are giving the same rating for each flight segment. If the data fall above the 45° line, males are rating more severely; and if the data fall below the line, females are rating more severely. The t-tests (see appendix B) were made at the 0.05 level of

significance to determine whether the data showed a statistically significant trend (or occurred by chance). From figure 12 and table V, it is seen that males rate the ride more severely than females throughout the range of vibration and noise levels for comfort ratings due to noise R_n and for overall comfort ratings R_0 . However, it should be noted that only a small sample of passenger-subjects was used in this study. These results are somewhat different from those described in reference 7 where it was shown that males rated the ride more severely than females in the rms vibration range below 0.04g, and then a reversal occurred in that females rated more severely than males at rms vibration levels greater than 0.055g. These latter results were also based on a small sample of passenger-subjects.

Effect of Flight Experience

The effect of flight experience is shown in figure 13 and table VI where the ratings by experienced subjects are compared with those of inexperienced subjects. The passenger-subjects with four or more flights were considered experienced. It is shown that experienced subjects rate more severely than inexperienced subjects for overall ratings $R_{\rm O}$. For ratings due to vibration and noise, no statistically significant difference was found at the 0.05 level.

Any further breakdown of subjects according to sex and flight experience makes for a very small sample size and, consequently, low reliability. Results will not be discussed here but can be found in appendix C, which includes a table indicating which comparisons have statistical significance at the 0.05 level.

Typical noise and vibration spectra are given in appendix D. For each condition of maximum thrust (including cruise), landing gear extended, and landing gear and speed brakes extended, both the acoustic and vibration spectra had the same shape, although their levels differed.

CONCLUDING REMARKS

Thirteen passenger-subjects were asked to rate their comfort associated with noise and vibration during a 1-hour flight aboard an NASA twin-jet airplane. The airplane was flown in a straight, horizontal path at a 3962-meter (13 000-foot) altitude. A range of vibration and noise levels was obtained by varying the thrust, the forward speed, and the position of the landing gear and drag brakes. The passenger-subjects' ages ranged from 18 to 48 years, weight from 54 to 84 kg (118 to 185 lb), and flying experience from first flight to many flights. Seven subjects were males and six were females; nine were professionals, three were students, and one was a homemaker.

For this flight, the noise and vibration levels were essentially independent of each other. For constant vibration level, good correlations were obtained between the vibration ratings and noise ratings, vibration ratings and noise levels, and noise ratings and noise levels. However, at constant noise levels, although the noise rating correlated well with vibration rating, the noise rating and the vibration rating both showed poor correlations with vibration level. Thus, during this flight passenger-subjects were able to distinguish and rate noise better than vibration.

Although only a small number of passenger-subjects were utilized, there was a statistically significant difference in ratings of ride comfort due to both sex type and experience in flying. Males rated flying discomfort much more severely than females when rating the overall ride and the ride when considering only the noise environment. Experienced passengers also rated the overall ride more severely than inexperienced passengers.

Langley Research Center
National Aeronautics and Space Administration
Hampton, VA 23665
August 2, 1976

APPENDIX A

RESULTS OF SECOND QUESTIONNAIRE SHOWN IN TABLE III

1.	Age: Weight: Height:	54 to 84 kg (118 1.6 m (5' 3") to 1.8 m	to 185 lb);		kg (137 lb)
2.	Sex:		Se	ven males; si	x females
3.	Occupation:	Nine professionals	; three st	udents; one ho	memaker
4.	Income:	From	less than	\$10,000 to ove	r \$30,000
5.	Purpose of trip:	92 per	cent busine	ess; 8 percent	pleasure
6.	Flight experience:	8 percent none; 23 per		1 to 3 previount greater than	
7.	Attitude toward flying:*	91 per	cent liked	it; 9 percent o	disliked it
8.	Importance of factors:*		None, percent	Moderate, percent	Very, percent
	(a) Trip cost		18	36	46
	(b) Time-saving		18	18	64
	(c) On-time arrival		18	55	27
	(d) Onboard services		46	46	8
	(e) Ride comfort		0	27	73
	(f) Convenience (door-to-do	oor)	27	56	27
	(g) Ability to read or write		18	64	18
9.	Time spent:*		None, percent	Some, percent	Most, percent
	(a) Reading	• • • • • • • • • • • • •	73	18	9
	(b) Writing		36	55	9
	(c) Talking		9	55	36
	(d) Looking out window		0	45	55
	(e) Dozing	• • • • • • • • • • • •	55	45	0
	(f) Thinking		9	73	18
	(g) Drinking or eating	· • • • • • • • • • • • • • • • • • • •	82	18	0
	*Based on respondents.				

APPENDIX A

10. Overall reaction to complete flight	10.	Overall	reaction	to	complete	flight
---	-----	---------	----------	----	----------	--------

Very comfortable, percent	. 8
Comfortable, percent	. 92
Neutral, percent	. 0
Uncomfortable, percent	. 0
Very uncomfortable, percent	. 0

11.	Feelings about environments:*	Not uncomfortable, percent	Somewhat uncomfortable, percent	Very uncomfortable, percent
*	(a) Lighting	100	0	0
	(b) Pressure (on ears)	67	33	0
	(c) Noise	25	67	8
	(d) Odors (other than			
	tobacco)	92	8	0
	(e) Presence of tobacco			
	smoke	92	8	0
	(f) Temperature	75	25	0
	(g) Ventilation	92	8	0
	(h) Workspace	92	8	0
	(i) General vibration	85	15	0
	(j) Sudden jolts	85	15	0
	(k) Bouncing	75	25	0
	(1) Back-and-forth motion	92	8	0
	(m) Side-to-side motion	85	15	0
	(n) Sudden descents	92	8	0
	(o) Turning	55	36	9
				Strongly

12.	Physical comforts of seat:*	Agree, percent	Disagree, percent	Strongly disagree, percent
	(a) Enough legroom	100	0	0
	(b) Satisfactory firmness	100	0	0
	(c) Wide enough	83	17	0
	(d) Satisfactory shape	83	17	0
	(e) Satisfactory adjustment	75	25	0

13. Specify seat location: see figure 2 of paper

^{*}Based on respondents.

APPENDIX A

14.	After this flight I would –	
	(a) Be eager to take another flight,* percent	83
	(b) Take another flight with confidence, percent	17
	(c) Take another flight, but with some doubt, percent	0
	(d) Perfer not to take another flight, percent	0
	(e) Not take another flight, percent	0
15.	Have you taken airsickness medication previously? 100 pe	rcent no
	This flight?	rcent no
	Did you experience any symptoms of airsickness on this flight? 100 pe	rcent no

*Based on respondents.

APPENDIX B

DEFINITIONS OF r AND t

The correlation coefficient r (see pp. 163 to 168 in ref. 9) gives an indication of the deviation of a set of data from a straight line. A value of 0 indicates no correlation or a complete deviation from a straight line. A value of ± 1 indicates that all samples fall on a straight line. Thus, r is defined as

$$\mathbf{r} = \frac{\sum_{i}^{n} (\mathbf{x}_{i} - \bar{\mathbf{x}})(\mathbf{y}_{i} - \bar{\mathbf{y}})}{{^{n}S_{\mathbf{X}}}^{S} \mathbf{Y}}$$

Omitting indexes and rewriting gives

$$\mathbf{r} = \frac{n \sum \mathbf{x} \mathbf{y} - \sum \mathbf{x} \sum \mathbf{y}}{\sqrt{\left[n \sum \mathbf{x}^2 - \left(\sum \mathbf{x}\right)^2\right] \left[n \sum \mathbf{y}^2 - \left(\sum \mathbf{y}\right)^2\right]}}$$

where x represents one set of data and y represents the other, S_X and S_Y are estimates of the standard deviations of the variables, and n is the number of observations.

The paired t-test is used to test an hypothesis that a sampling of data occurred because of a treatment rather than by chance, particularly when extraneous factors cause a significant difference in means. A t-value is calculated and compared to a value in a set of tables based on degrees of freedom and confidence levels. Since any effect, either positive or negative, would be accepted, a two-tailed t-value is compared to the calculation. Thus, for the data in this paper, for 12 degrees of freedom (13 treatments) at the 0.05 level of significance, the value of t from the table in reference 9 (p. 393) is 2.179. If the calculated value of t is greater than 2.179, then the data show a significant change due to the treatment. Thus, t is calculated as follows (see ref. 10, pp. 115 to 127):

$$t = \frac{\bar{d}}{s^2/\sqrt{N}}$$

where $\bar{d} = \bar{X}_1 - \bar{X}_2$ is the difference of the means of the populations, s^2 is the variance of the differences, and N is the number of paired observations.

APPENDIX C

EFFECT OF SEX TYPE AND FLIGHT EXPERIENCE

In this appendix the following tables and figures show additional effects of sex type and experience in flying as a passenger in an airplane. In table CI showing paired ratings, the averages of the males and females with and without flight experience are given. In table CII showing statistical significance, the computed "t-values" are shown as well as whether the data plotted in figures C1 and C2 show statistical significance at the 0.05 level. In these figures, a flagged symbol represents a duplicate data point. Although these data tend to confirm the conclusions pertaining to sex type and experience, the data sample is so small that no conclusions should be drawn from these data.

APPENDIX C

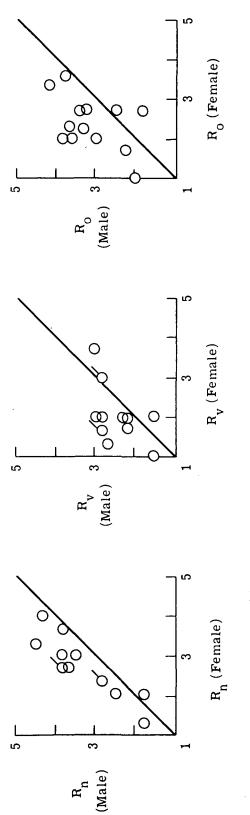
TABLE CI.- TABLE OF PAIRED RATINGS

	Expe	Experience	Inexp	Inexperience	Expe	Experience	Inexpe	Inexperience	Expe	Experience	Inexp	Inexperience
rms vertical vibration, g units	Males	Females	Males	Females	Males	Females	Males	Females	Males	Females	Males	Females
		Noise rating,	ing, R _n			Vibration rating,	ating, R _v	Λ		Overall rating,	ating, R _o	
0.0285	3.7	2.7	2.0	2.3	2.2	2.0	2.0	2.3	3.0	2.0	2.0	2.0
.0312	2.5	2.0	2.0	2.3	2.5	1.3	1.0	2.0	2.2	1.7	2.0	1.7
.0316	3.8	2.7	3.0	3.3	2.8	2.0	2.0	2.7	3.6	2.0	3.0	2.3
.0319	3.8	2.7	3.0	3.7	2.8	1.7	2.0	3.0	3.7	2.3	2.0	2.7
.0403	4.3	3.0	3.0	3.7	3.0	2.0	2.0	2.7	3.8	2.0	3.0	2.0
_	2.8	2.3	3.0	2.3	2.8	3.0	2.0	2.0	3.2	2.7	3.0	3.0
&	3.8	3.7	3.0	4.0	2.8	2.7	2.0	3.3	3.4	2.7	2.0	3.3
.0486	1.8	1.3	2.0	1.3	1.5	1.0	1.0	1.3	2.0	1.0	2.0	2.0
.0506	2.8	2.3	1.0	2.7	2.3	2.0	1.0	2.0	2.5	2.7	2.0	2.7
.0536	4.5	3.3	4.0	4.3	2.8	3.0	2.0	2.3	3.7	3.6	3.0	3.3
.0549	3.8	3.0	4.0	4.0	3.0	3.7	3.0	3.3	4.2	3.3	4.0	3.7
1029	1.8	2.0	2.0	1.3	1.6	2.0	1.0	1.0	1.8	2.7	2.0	1.7
1161	3.5	3.0	3.0	2.7	2.2	1.7	2.0	1.7	3.3	2.3	2.0	2.7

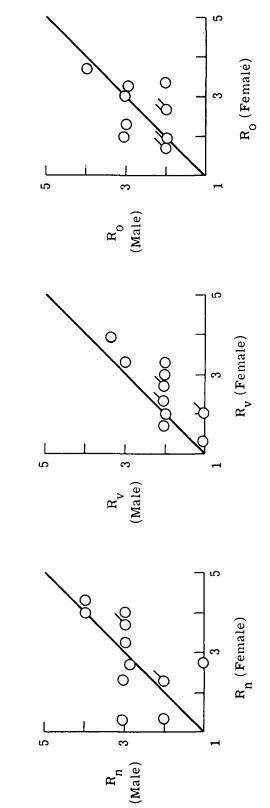
APPENDIX C

Chabitatian and making	t-	Sign	ificance
Statistical rating	value	Yes	No
Noise rating			
Exp. males to inexp. males	-3.14	/	
Exp. females to inexp. females	1.93	v	.,
Exp. males to exp. females	-5.52	_/	*
Inexp. males to inexp. females	1.12	•	/
Inexp. males to exper. females	.42		/
Exper. males to inexp. females	-3.23	√	,
Vibration rating	·		<u></u>
Exp. males to inexp. males	-5.94	✓	
Exp. females to inexp. females	.58		✓ .
Exp. males to exp. females	-1.87		✓
Inexp. males to inexp. females	3.81	✓	
Inexp. males to exper. females	-2.71	✓	
Exper. males to inexp. females	-1.89		✓
Overall rating	·		
Exp. males to inexp. males	-4.03	√	
Exp. females to inexp. females	1.21		✓
Exp. males to exp. females	-3.51	\checkmark	
Inexp. males to inexp. females	.48		✓
Inexp. males to exper. females	41	ļ	✓ .
Exper. males to inexp. fémales	-3.53	✓	

^{*}See figures C1 and C2.



(a) Experienced passengers with four or more flights.



(b) Inexperienced passengers with fewer than four flights.

Figure C1.- Comparisons of male and female ratings. Each flag on a symbol represents a duplicate data point.

15

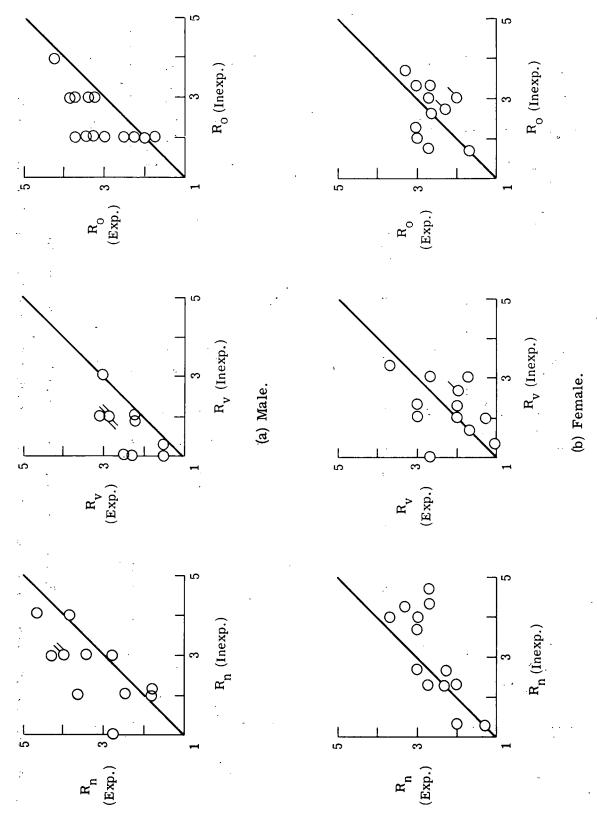
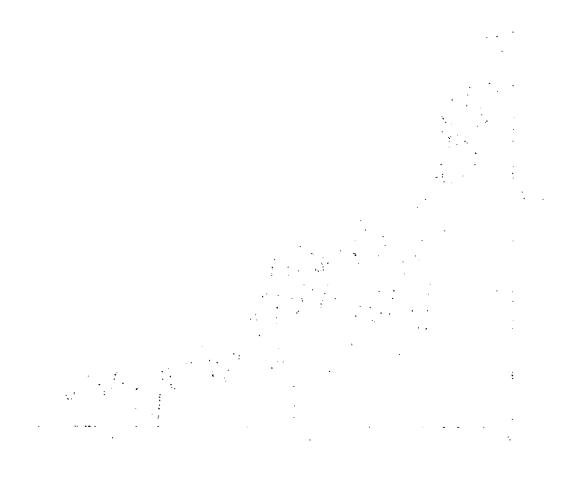


Figure C2.- Comparisons of subject ratings with and without flight experience. Each flag on a symbol represents a duplicate data point.

APPENDIX D

TYPICAL VIBRATION AND NOISE SPECTRA

In this appendix figure D1 shows typical vibration spectra for cruise and maximum cruise, landing gear extended, and landing gear and speed brakes extended. Each curve is representative of the flight condition shown, with all having similar shapes but different amplitudes. Figure D2 shows typical unweighted sound-pressure level conditions. Although the three spectra are not alike, the repetitions of each flight segment did result in similar spectra, although at somewhat different levels.



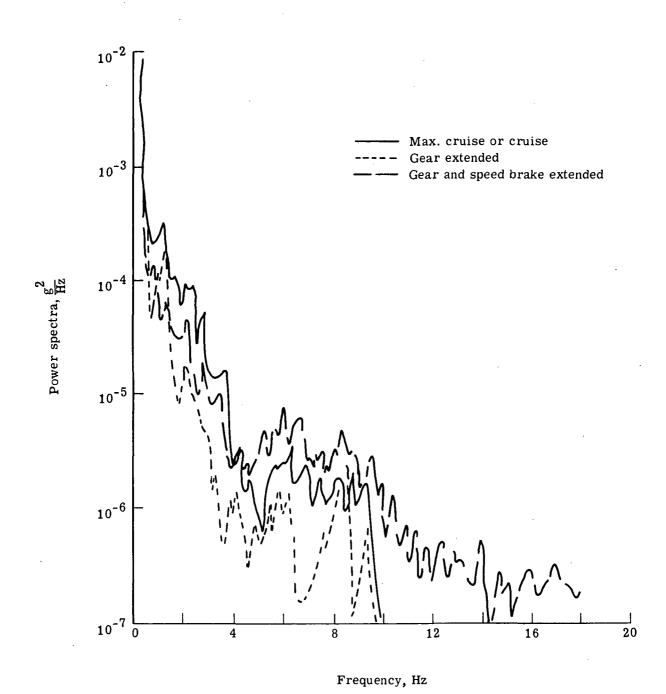


Figure D1.- Typical vibration spectra.

APPENDIX D

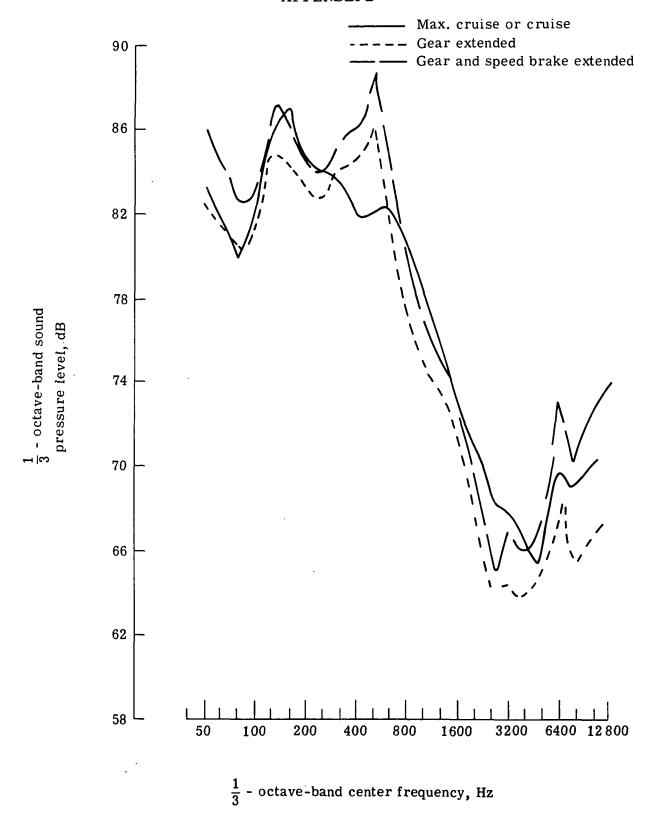


Figure D2. - Typical noise spectra.

REFERENCES

- 1. Gruesbeck, Marta G.; and Sullivan, Daniel F.: Aircraft Motion and Passenger Comfort Data From Scheduled Commercial Airline Flights. NASA CR-2612, 1976.
- 2. Jacobson, Ira D.; Schoultz, Michael B.; and Blake, J. Coleman: Effect of Motion Frequency Spectrum on Subjective Comfort Response. Rep. 403901, Transp. Div., Univ. of Virginia, Nov. 1973. (Available as NASA CR-138883.)
- 3. Stone, Ralph W., Jr.: Ride Quality An Exploratory Simulator Study and Criteria Development. NASA TM X-71922, 1974.
- 4. Dempsey, Thomas K.; and Leatherwood, Jack D.: Experimental Studies for Determining Human Discomfort Response to Vertical Sinusoidal Vibration. NASA TN D-8041, 1975.
- 5. Guignard, J. C.; and King, P. F.: Aeromedical Aspects of Vibration and Noise. AGARD AG-151, Nov. 1972.
- 6. 1975 Ride Quality Symposium. NASA TM X-3295, DOT-TSC-OST-75-40, 1975.
- 7. Duncan, Nancy C.; and Conley, Harold W.: Demographic and Psychological Variables Affecting Test Subject Evaluations of Ride Quality. 1975 Ride Quality Symposium, NASA TM X-3295, DOT-TSC-OST-75-40, 1975, pp. 287-321.
- 8. Catherines, John J.; Clevenson, Sherman A.; and Scholl, Harland F.: A Method for the Measurement and Analysis of Ride Vibrations of Transportation Systems. NASA TN D-6785, 1972.
- 9. Hoel, Paul G.: Introduction to Mathematical Statistics. Fourth Ed., John Wiley & Sons, Inc., c.1971.
- 10. Dixon, Wilfrid J.; and Massey, Frank J., Jr.: Introduction to Statistical Analysis. Second Ed., McGraw-Hill Book Co., Inc., 1957.

TABLE I.- CHARACTERISTICS OF THE NASA TWIN-JET AIRPLANE

Maximum take-off weight, kg (lb)	(97 800)
Maximum landing weight, kg (lb)	(89 700)
Maximum zero full weight, kg (lb)	(85 000)
Operating empty weight, kg (lb)	(68 000)
Wing span, m (ft)	(93)
Overall length, m (ft)	(94)
Overall height, m (ft)	(37)

TABLE II.- QUESTIONNAIRE # 1

Segment #:	egment by checking one of	Acceptable	dividual Features of the pre-
Flight #:	indicate your overall reaction to the previous flight segment by checking one of	Comfortable Ac	page below, we would like you to rate the following individual Features of the pre-
FI	Please indicate your overall roof the descriptions below.	Very comfortable Uncomfortable	On the page below, we would li
Subject #:	jo		

ceding ride segment. Along a five point scale ranging from "comfortable" to "uncomfortable," judgement - - - conversely, the higher the number you circle, the more "uncomfortable" your you will circle number 1, 2, 3, 4, or 5; the lower the number, the more "comfortable" your judgement of that item.

	Comfortable					Uncomfortable
Noise		.23	က	4	2	
	Comfortable					Uncomfortable
General vibration	1	87	က	4	ល	

TABLE III.- QUESTIONNAIRE #2

Subject #:			Flight #:	Segment #:
نټ	The objective hat future transpor	ive of this questior iportation systems	The objective of this questionnaire is to identify the needs and desthat future transportation systems may increase passenger satisfaction.	of this questionnaire is to identify the needs and desires of passengers so tation systems may increase passenger satisfaction.
F	1. Age		Weightlbs	Height "
87	2. Sex:	M O	OF	
ന	3. Primary occupation:	upation: (Check one)	ne)	
	О Но	O Homemaker		O Craftsman, Mechanic
	O Student	dent		O Secretary, Clerical
	O Sales	es		O Professional
	O Ma	O Manager, Official, Executive	ecutive	O Other (specify)
4	4. Approximate	total household in	Approximate total household income (before taxes):	
	O Unc	O Under \$10,000		O \$20,000-30,000
	O \$10	O \$10,000-20,000		O \$30,000 or more
വ	5. What is the I	What is the primary purpose of this trip?	this trip? (Check one)	
		O Company business O Derechal business		
	O Ple	O Pleasure		
9	About bour	our times have	About how many times have you traveled by this mode of transportation?	tranenortation?
J		O (This is my first trip.)	p.)	
	O 1-3			
	O 4 0	O 4 or more		

TABLE III. - Continued

transportation?
jo
mode (
this
S,
traveling
l about
feel
yon
ဝ
How do you feel
7.

- I like it.
- O I have no strong feelings.
- O I dislike it.
- I use this mode of transportation because I have to.
- 8. Indicate the importance of each of the following factors in your choice of way to travel (this trip).

		ĺ			
lime-saving	In-time arrival and departure	Inboard services	tide comfort	Convenience (door-to-door)	Ability to read or write
		al and departure O O O	nd departure O O O O O O O O O O O O O O O O O O O	al and departure O O O O O O O O O O O O O O O O O O O	Time-saving O <td< td=""></td<>

9. Check the box which indicates how much time during this trip you spent doing each of the following:

Some Most	0 0						ŀ
None	0	0) C		0	erform?
	Reading	Writing Talking	Looking out the window	Dozing	Thinking	Drinking or eating	Were any of the above difficult to perform?

0. F	Please indicate your overall reaction to this flight. (Check one.):
	O Very comfortable
	O Comfortable
	O Neutral
	O Uncomfortable
	O Very uncomfortable
	Comment
1. C	Check the box which indicates your feelings about each of the following items on this flight.
) 20
	Stor Stan
	those in those in those in the second in the
1	olde Inde
14	e (on ears)
4	
O	
щ	
[-	
-	
<i>i</i> >	
J	
Ø	
ב	
щ	
02	
02	
ר	Q
O	

TABLE III. - Concluded

eat.
t your s
about
atements
lowing st
the follo
each of
on to
our reaction to each
e your
Indicat
12.

		OO TO	88.46 \$1810
It has enough leg room.	top (\\\\\\\\\\\\\\\\\\\\\\\\\\\\\\\\\\\\\\
ory.			
It is wide enough.			
Its shape is satisfactory.			
Its adjustment is satisfactory.			
Specify your seat location:	ı	ı	ı
(Check one)		(Check the firs	(Check the first which applies.)
O Ahead of the wing		O Window seat	

location:
ur seat
ecify your
13. Spe

	O No	O Yes	Was an adjacent seat occupied?
O Aisle seat			O Behind the wing
O Center seat			O Over or under the wing
O Window seat			O Ahead of the wing
(Check the first			(Check one)

14. After experiencing this flight, I would:

(Check only one.)

- O be eager to take another flight
- O take another flight with confidence
- O take another flight, but with some doubt
 - O prefer not to take another flight O not take another flight

		O No
O No	O No	O Yes
O Yes	O Yes	ight?
Previously	This flight	ckness on this fl
15. Have you taken airsickness medication?		Did you experience any symptoms of airsickness on this flight?
13		

16. General comments

TABLE IV.- NOISE AND VIBRATION LEVELS
AND ASSOCIATED SUBJECTIVE RATINGS

	bration g units	Noise	·	Average rati	ngs
Vertical	Lateral*	level, dB(A)	Noise, R _n	Vibration, R _V	Overall, R _O
0.0285	0.0056	86	3.00	2.15	2.46
.0312	.0102	93	2.31	2.00	1.92
.0316	.0087	86	3.33	2.50	2.75
.0319	.0075	91	3.46	2.54	2.85
.0403	.0064	95	3.64	2.55	2.73
.0447	.0076	88	2.62	2.62	3.07
.0448	.0083	90	3.77	2.82	2.92
.0486	.0065	81	1.62	1.31	1.77
.0506	.0054	87	2.54	2.08	2.54
.0536	.0073	94	4.15	2.69	3.53
.0549	.0074	94	3.69	3.23	3.85
.1029	.0091	84	1.77	1.39	1.84
.1161	.0106	94	3.15	1.92	2.85

^{*}Included for reference only (not referred to elsewhere).

TABLE V.- COMPARISON OF MALE AND FEMALE RATINGS

rms vibration, g units	Noise rating, R _n		Vibration rating, R _V		Overall rating,	
	Male	Female	Male	Female	Male	Female
0.0285	3.4	2.5	2.1	2.2	2.8	2.0
.0312	2.4	2.2	. 2.3		2.1	1.7
.0316	3.7	3.0	2.7	2.3	3.5	2.2
.0319	3.7	3.2	2.7	2.3	3.3	2.3
.0403	4.0	3.3	2.8	2.3	3.6	2.0
.0447	2.9	2.3	2.7	2.5	3.1	3.0
.0448	3.6	3.8	2.7	3.0	3.0	2.8
.0486	1.9	1.3	1.4	1.2	2.0	1.5
.0506	2.6	2.5	2.1	2.0	2.4	2.7
.0536	4.4	3.8	2.6	2.7	3.6:	3.5
.0549	3.9	3.5	3.0	3.5	4.1	, 3.5
.1029	1.9	1.7	1.6	1.3	1.9	1.8
.1161	3.4	2.8	2.1	1.7	2.9	2.5
^t critical		2.179		2.179	;	2.179
Paired t	r	5.42		1.78	,	3.48
Statistically significant		Yes		No	·	Yes

TABLE VI.- COMPARISON OF RATINGS DUE TO EXPERIENCE

rms vibration, g units	Noise rating, R _n		Vibration rating, R _V		Overall rating, R _O	
	Exp.	Inexp.	Exp.	Inexp.	Exp.	Inexp.
0.0285	3.3	2.3	2.1	2.3	2.7	2.0
.0312	2.3	2.3	2.1	1.8	2.2	1.8
.0316	3.4	3.3	2.5	2.5	2.9	2.5
.0319	3.4	3.5	2.4	2.8	3.2	2.5
.0403	3.9	3.5	2.6	2.5	3.0	2.3
.0447	2.7	2.5	2.9	2.0	3.1	3.0
.0448	3.7	3.8	2.8	3.0	3.1	2.8
.0486	1.8	1.5	1.3	1.3	1.7	2.0
.0506	2.7	2.3	2.2	1.8	2.6	2.5
.0536	4.1	4.2	2.9	2.3	3.6	3.3
.0549	3.6	4.0	3.2	3.3	3.9	3.8
.1029	1.9	1.5	1.7	1.0	2.1	1.8
.1161	3.3	2.8	2.0	1.8	3.0	2.5
^t critical		2.179		2.179		2.179
Paired t	2.01		1.65		4.12	
Statistically significant		No		No		Yes

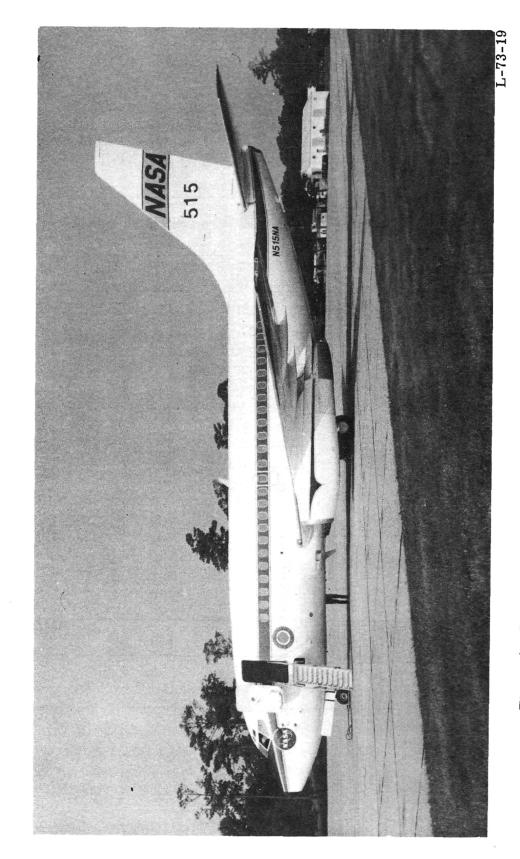


Figure 1.- Large twin-jet airplane used in flight study of subjective response.

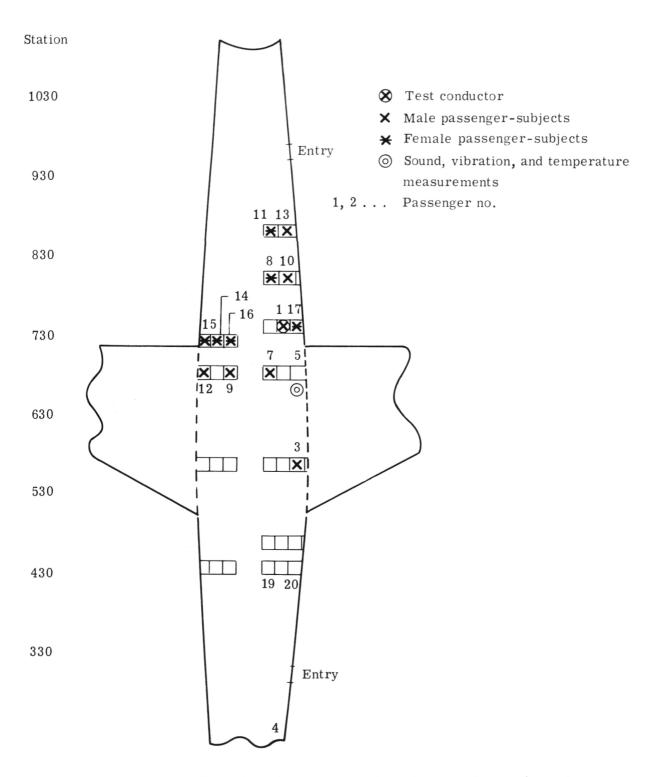


Figure 2.- Location of passenger-subjects and measuring equipment.

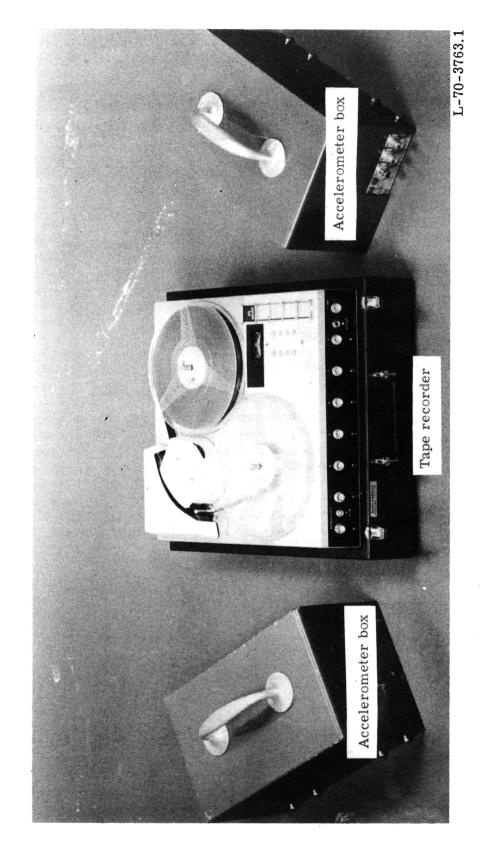


Figure 3.- Equipment for measuring and recording vibration.

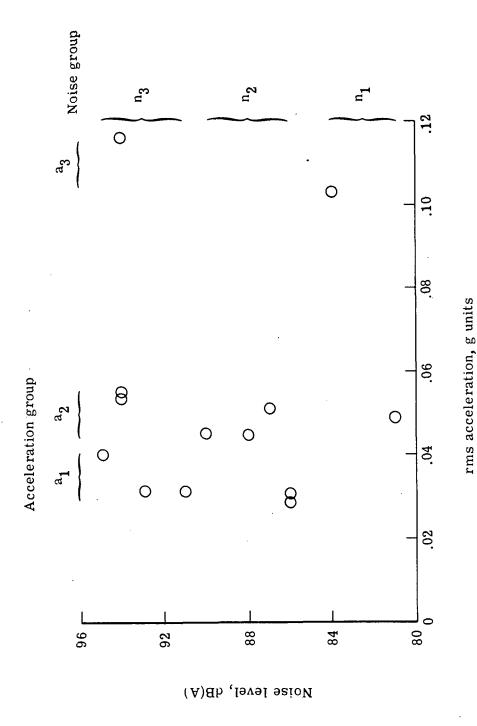


Figure 4.- Ranges of vertical (<30 Hz) vibration and noise levels. r = 0.038.

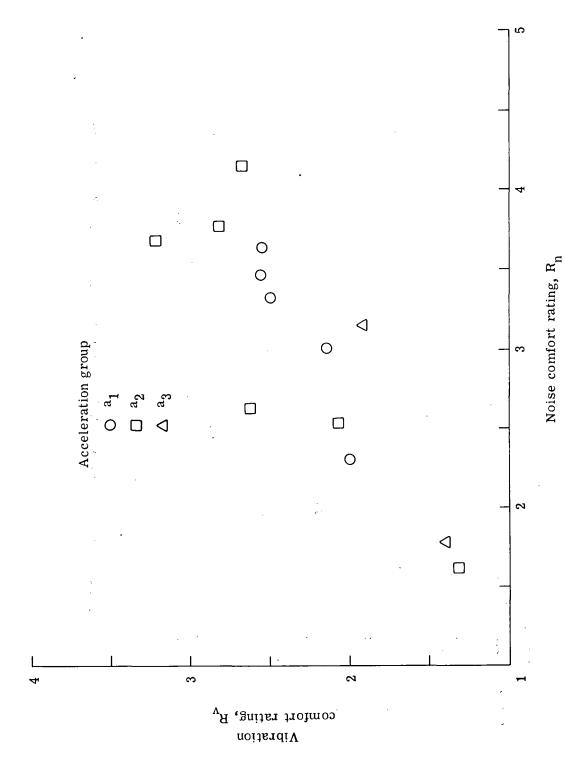
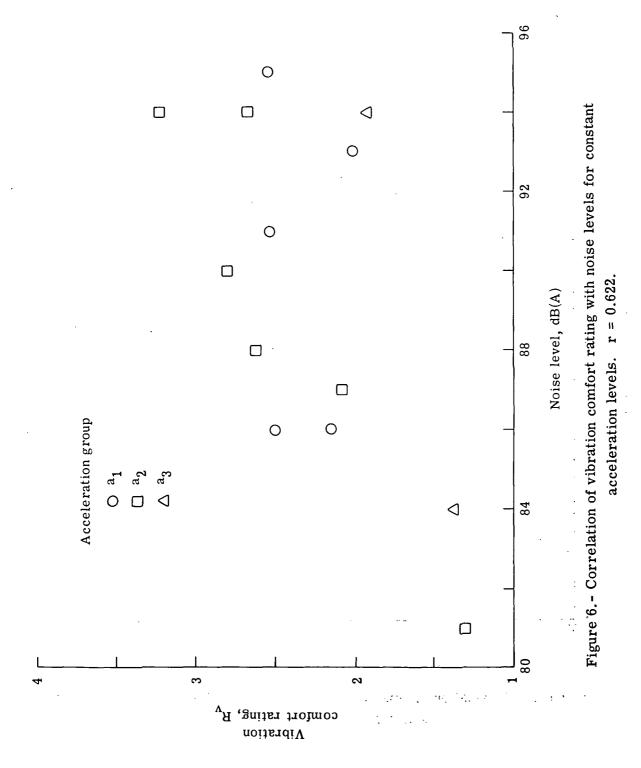


Figure 5.- Correlation of vibration comfort rating with noise comfort rating for constant acceleration levels. r = 0.854.



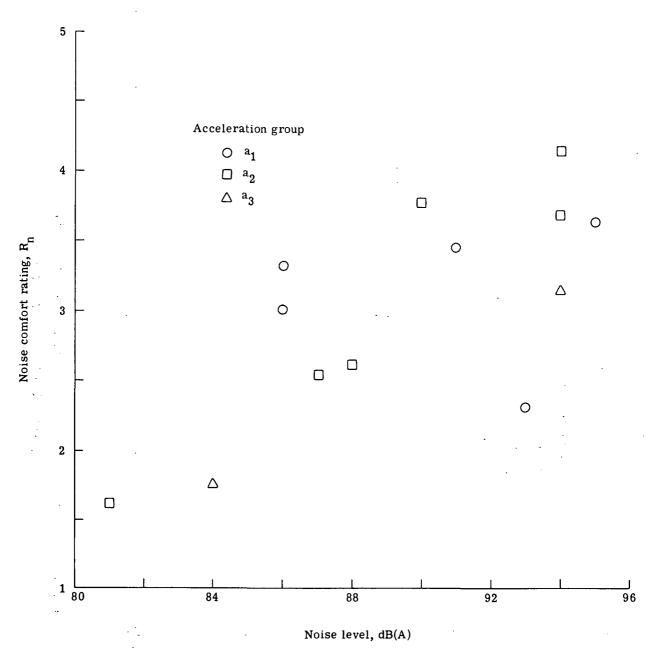


Figure 7.- Correlation of noise comfort rating with noise levels for constant acceleration levels. r = 0.715.

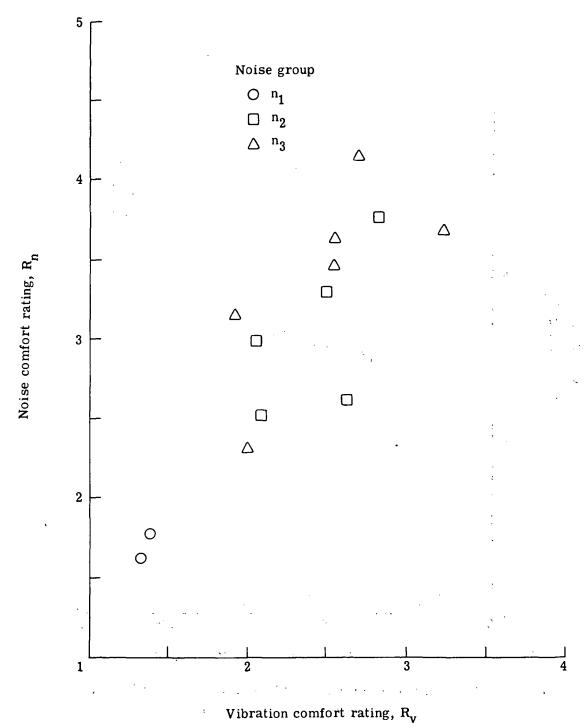


Figure 8.- Correlation of noise comfort rating with vibration comfort rating for constant noise levels. r = 0.854.

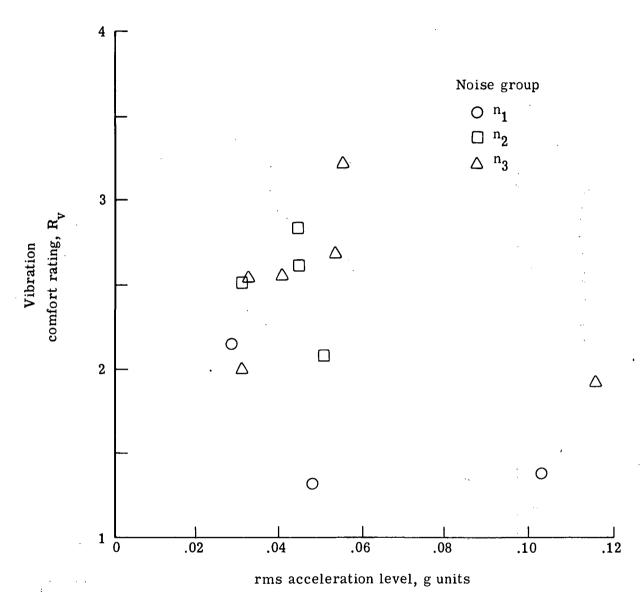
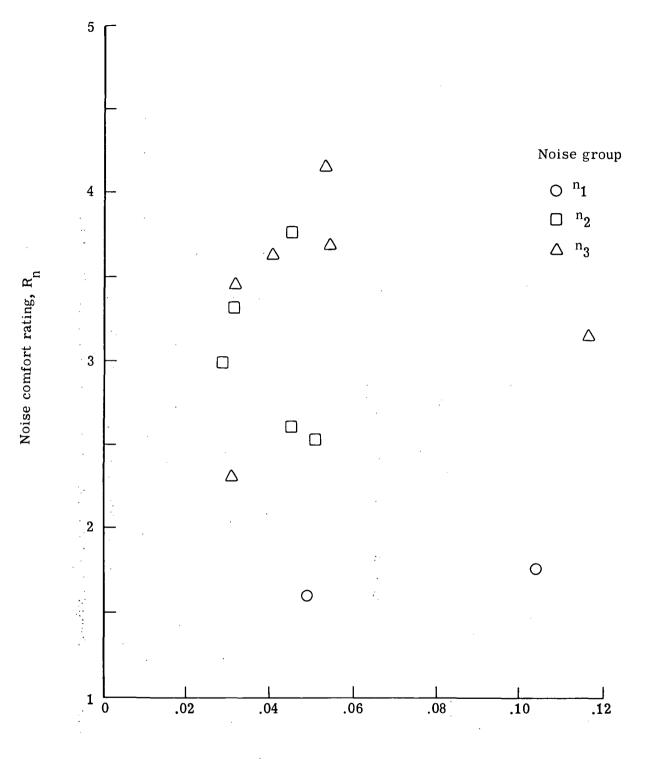


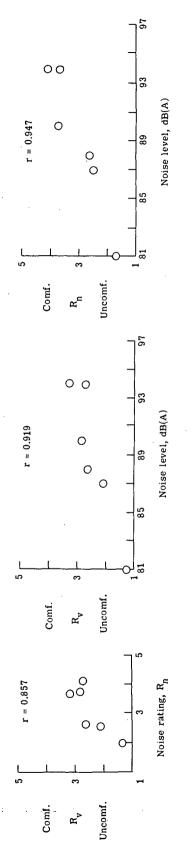
Figure 9.- Correlation of vibration comfort rating with acceleration level for constant noise levels. r = -0.401.

38

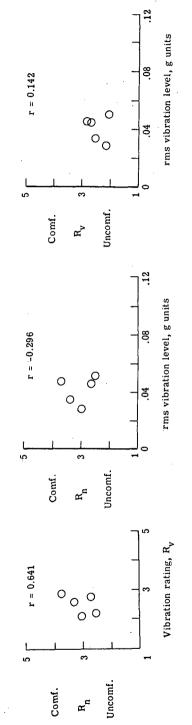


rms acceleration level, g units

Figure 10.- Correlation of noise comfort rating with acceleration level for constant noise levels. r = -0.219.



(a) Constant vibration level. $g_{rms} = 0.0498 \pm 0.0051$.



(b) Constant noise level. $dB(A) = 88 \pm 2$.

Figure 11.- Composite correlations of noise and vibration ratings with environments.

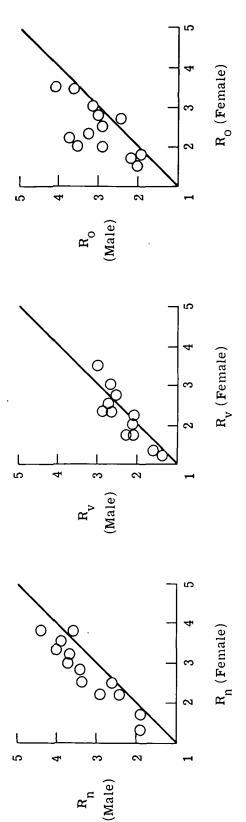


Figure 12.- Comparison of ride comfort ratings by male and female passengers.

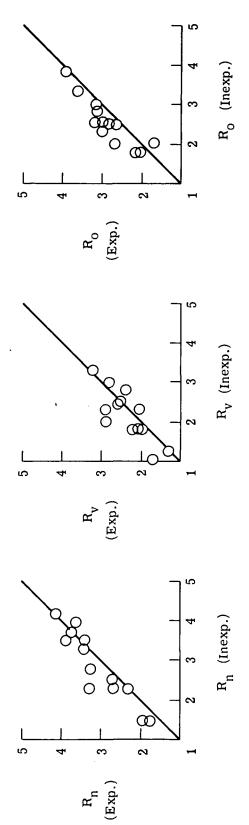


Figure 13.- Comparison of ride comfort ratings by experienced and inexperienced passengers.

OFFICIAL BUSINESS
PENALTY FOR PRIVATE USE \$300

SPECIAL FOURTH-CLASS RATE BOOK



POSTMASTER:

If Undeliverable (Section 158 Postal Manual) Do Not Return

"The aeronautical and space activities of the United States shall be conducted so as to contribute . . . to the expansion of human knowledge of phenomena in the atmosphere and space. The Administration shall provide for the widest practicable and appropriate dissemination of information concerning its activities and the results thereof."

-NATIONAL AERONAUTICS AND SPACE ACT OF 1958

NASA SCIENTIFIC AND TECHNICAL PUBLICATIONS

TECHNICAL REPORTS: Scientific and technical information considered important, complete, and a lasting contribution to existing knowledge.

TECHNICAL NOTES: Information less broad in scope but nevertheless of importance as a contribution to existing knowledge.

TECHNICAL MEMORANDUMS:

Information receiving limited distribution because of preliminary data, security classification, or other reasons. Also includes conference proceedings with either limited or unlimited distribution.

CONTRACTOR REPORTS: Scientific and technical information generated under a NASA contract or grant and considered an important contribution to existing knowledge.

TECHNICAL TRANSLATIONS: Information published in a foreign language considered to merit NASA distribution in English.

SPECIAL PUBLICATIONS: Information derived from or of value to NASA activities. Publications include final reports of major projects, monographs, data compilations, handbooks, sourcebooks, and special bibliographies.

TECHNOLOGY UTILIZATION
PUBLICATIONS: Information on technology used by NASA that may be of particular interest in commercial and other non-aerospace applications. Publications include Tech Briefs, Technology Utilization Reports and Technology Surveys.

Details on the availability of these publications may be obtained from:

SCIENTIFIC AND TECHNICAL INFORMATION OFFICE

NATIONAL AERONAUTICS AND SPACE ADMINISTRATION

Washington, D.C. 20546